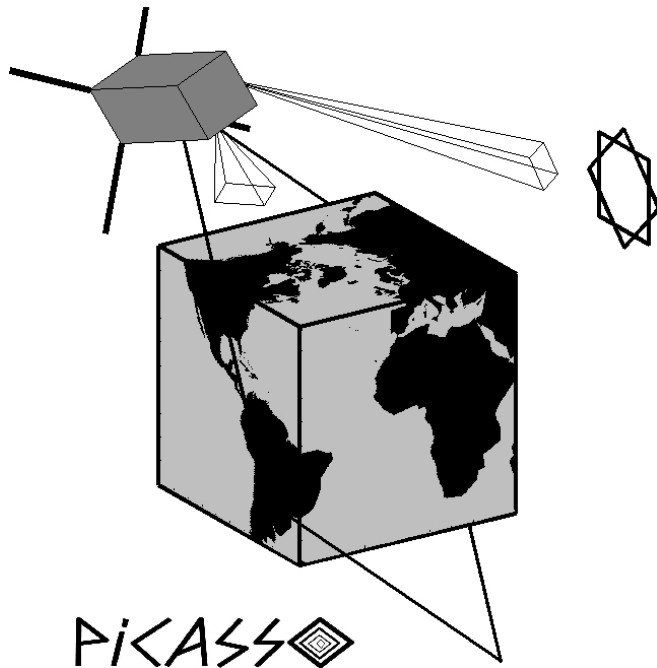




Use of a Langmuir Probe Instrument on Board a Pico-Satellite



Sylvain Ranvier¹, Michel Anciaux¹, Pepijn Cardoen¹,
Sabrina Bonnewijn¹, Emmanuel Gamby¹, Johan De
Keyser¹, Didier Pieroux¹, Jean-Pierre Lebreton²

¹ Royal Belgian Institute for Space Aeronomy (BIRA-IASB)

² Laboratoire de Physique et Chimie de l'Environnement
et de l'Espace (LPC2E)

Contact: sylvain.ranvier@aeronomie.be

1. PICASSO mission
2. PICASSO platform
3. Expected plasma environment
4. SLP instrument
5. LP on board pico-satellite
6. Proposed solution
7. Simulations
8. Probe potential for extreme cases
9. Conclusions

- ESA in-orbit demonstrator
- Quasi polar orbit, altitude: 400 - 700 km
- expected orbital lifetime: 1-2 years
- Launch in 2017
- 2 scientific instruments:

VISION: visible and near-infrared hyper-spectral imager

Scientific objectives:

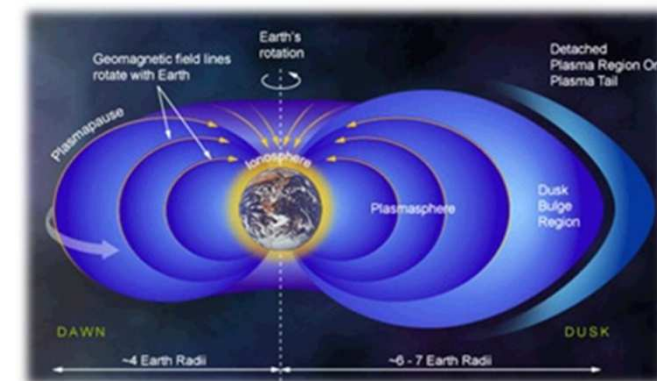
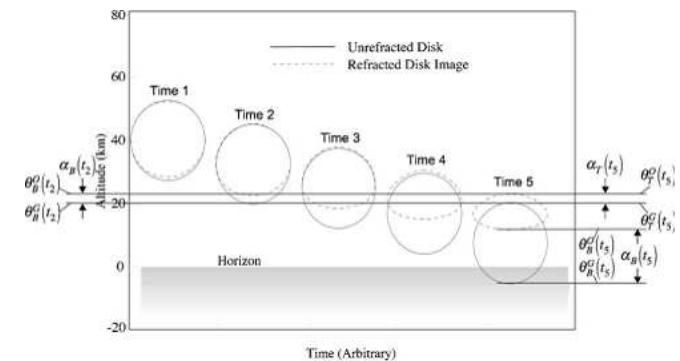
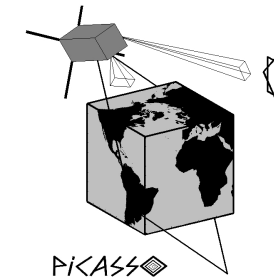
1. Polar and mid-latitude stratospheric ozone vertical profile retrieval
2. Upper atmosphere temperature profiling based on the Sun refractive flattening

SLP: Sweeping Langmuir Probe

Scientific objectives: in-situ study of

1. Ionosphere-plasmasphere coupling
2. Subauroral ionosphere and corresponding magnetospheric features
3. Aurora structures
4. Turbulence (multi-scale behavior, spectral properties)

PICASSO mission





PICASSO platform

Triple unit (34x10x10 cm, 1U for payload), four 2U deployable solar panels

Telecom: UHF/VHF + S-Band

Attitude control: magneto-torquers & dynamical wheels

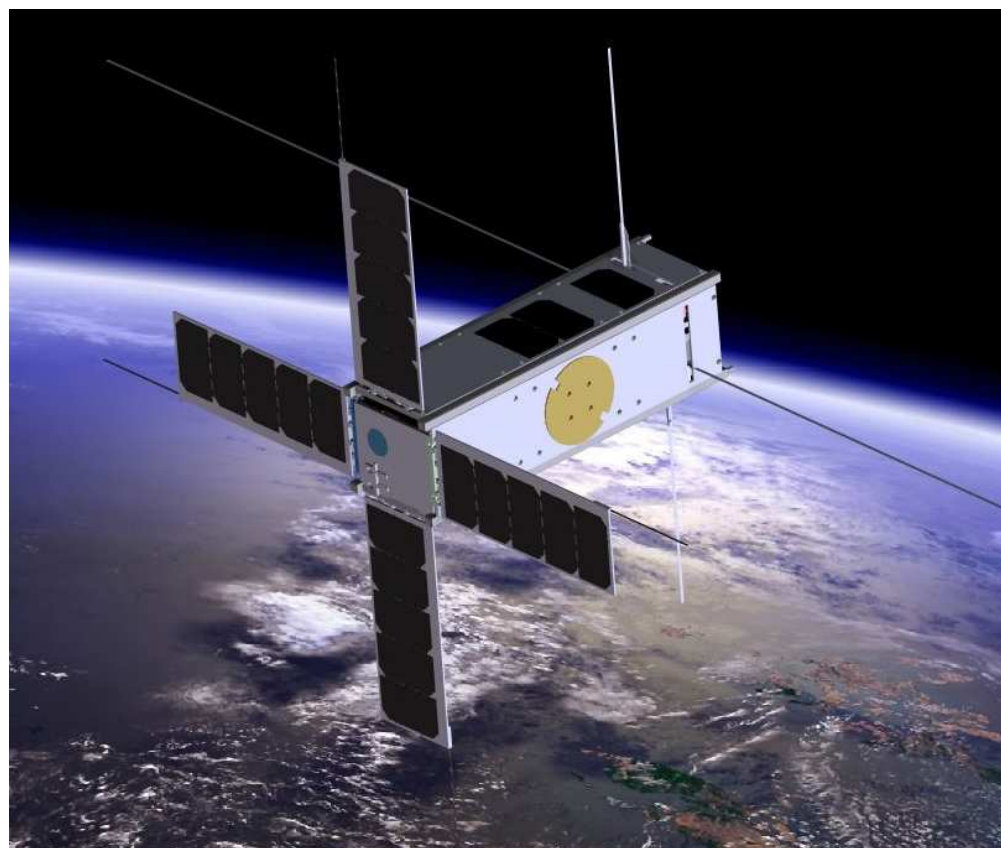
Attitude: inertial flight, one face towards the Sun

Mass: 3.9 kg, power consumption: 6.3 W

Downlink: 40 MB/day, Uplink: 400 kB/day

Pointing accuracy: $\sim 1^\circ$ (knowledge: 0.2°)

Star tracker & GPS





Expected plasma environment

	Minimum (> 95% probability)	Maximum (> 95% probability)
Plasma density (#/m ³)	10 ⁸ (10 ⁹)	10 ¹³ (5x10 ¹²)
Electron temperature (K)	600 (700)	10 000 (5 000)
Debye length (m)	5.4e-4 (8.2e-4)	0.69 (0.15)
Electron plasma frequency (Hz)	5.7e5 (1.8e6)	1.8e8 (1.3e8)



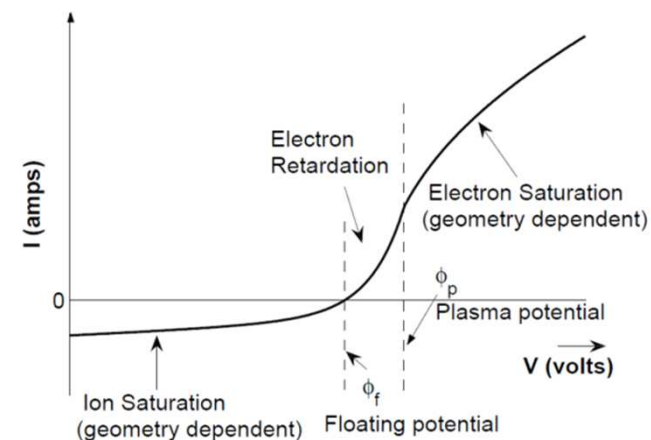
SLP instrument

Instrument overview

SLP is made of 4 thin cylindrical Langmuir probes whose electrical potentials are swept with respect to the S/C potential. From the electric current collected by each probe, the following parameters will be retrieved on ground:

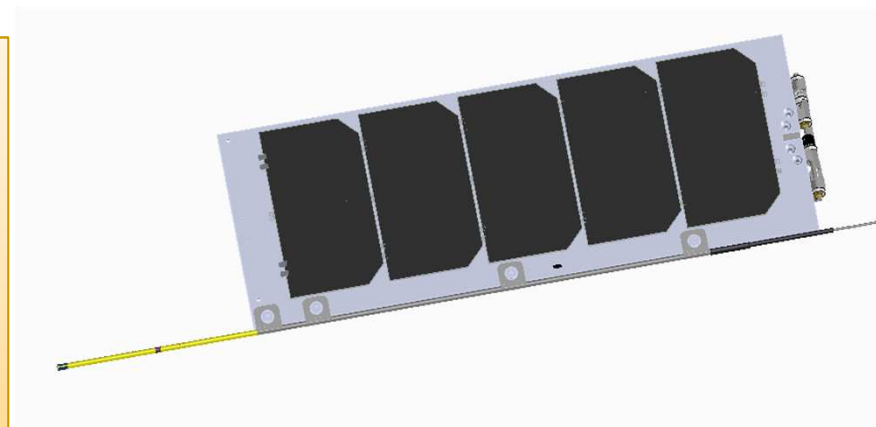
- ✓ electron density and temperature
- ✓ ion density
- ✓ Spacecraft potential

Download raw data: I-V curves contain more information than only 3 parameters !



Key numbers:

- 4 probes
- Probe diameter: 2 mm
- Probe length: 40 mm
- Boom length: 40 mm
- Sampling frequency: 10 KHz
- 50 I-V curves/s \Rightarrow Ne, Te, Ni, $V_{S/C}$
- Bias voltage: -5 V to 13 V wrt S/C GND





LP on board Pico-Satellite

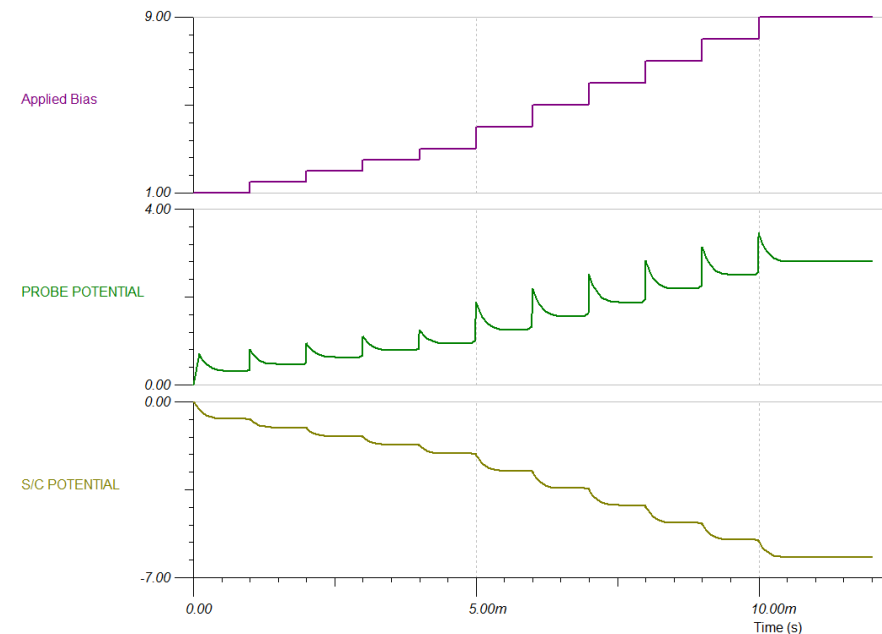
Problem of using LP on board Pico-Satellite

Limited conducting area of the S/C with respect to the area of the probe

- ⇒ **Spacecraft charging** (e- saturation region)
 - ⇒ Drift of the instrument's electrical ground during the measurement
 - ⇒ Unusable data

Risk:

Too low S/C potential: unable to sweep appropriate potentials (e- saturation region)





Proposed solution

Proposed solution

- Increase conducting surface of the S/C (at least 200 cm² on all sides of the S/C, incl. solar panels)
- Measure the floating potential of one probe while measuring the I-V curve with another probe
=> The 2 probes that are in the same environment (light/shadow, wake)

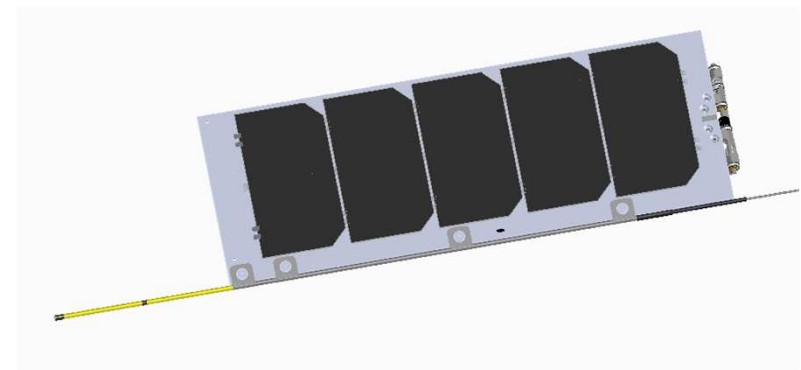
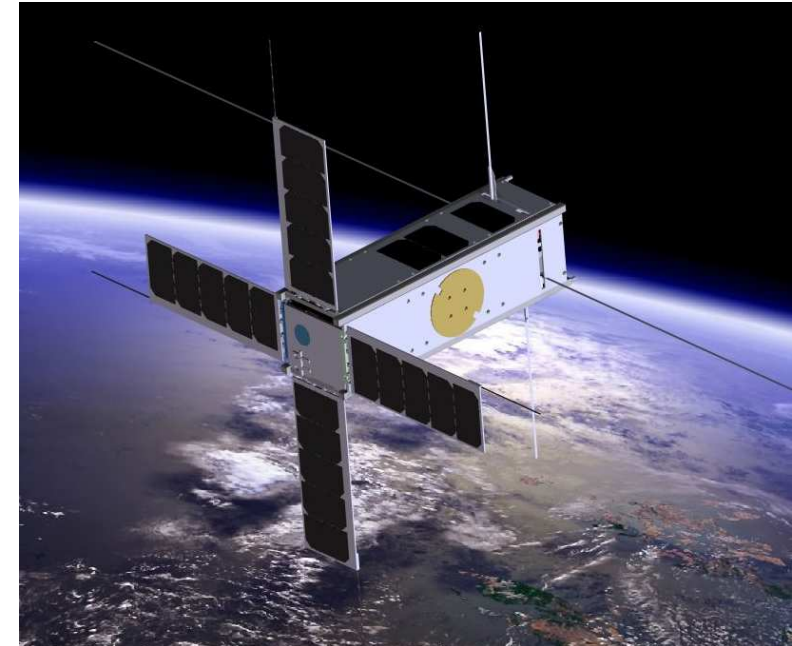
Advantages

Robust: no filament

No risk of electron collection from e-gun

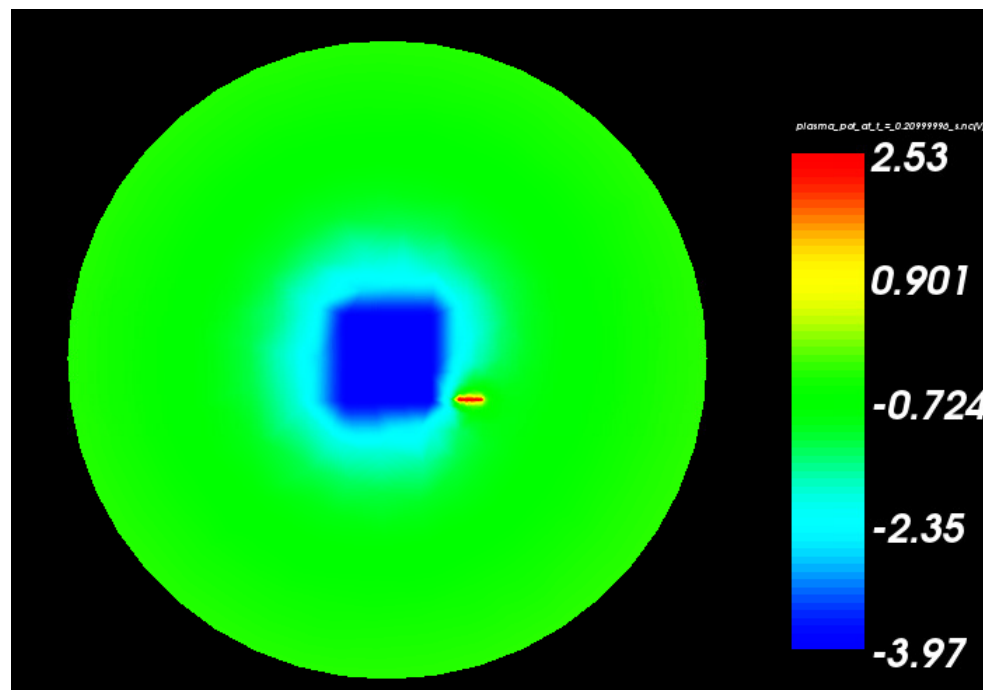
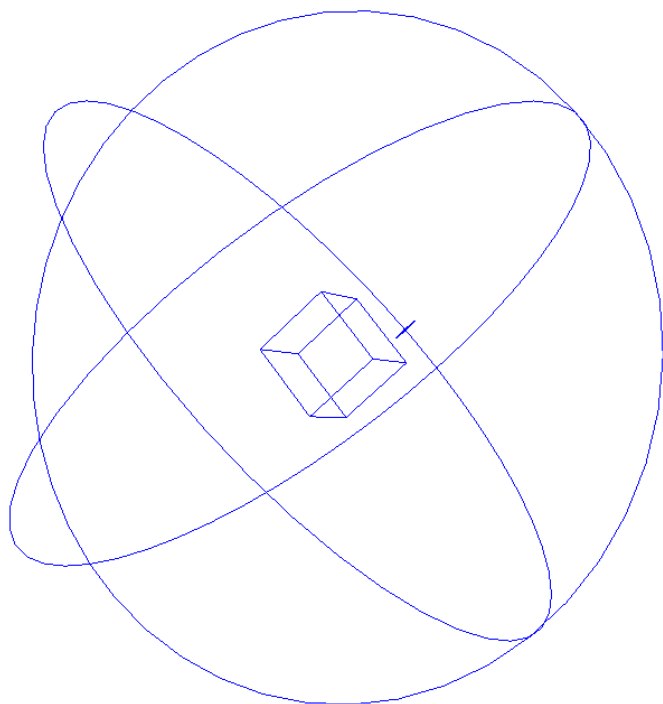
Disadvantage

Limited range in e- saturation region in very high density plasma





Particle-in-cell (PIC) modelling and simulations

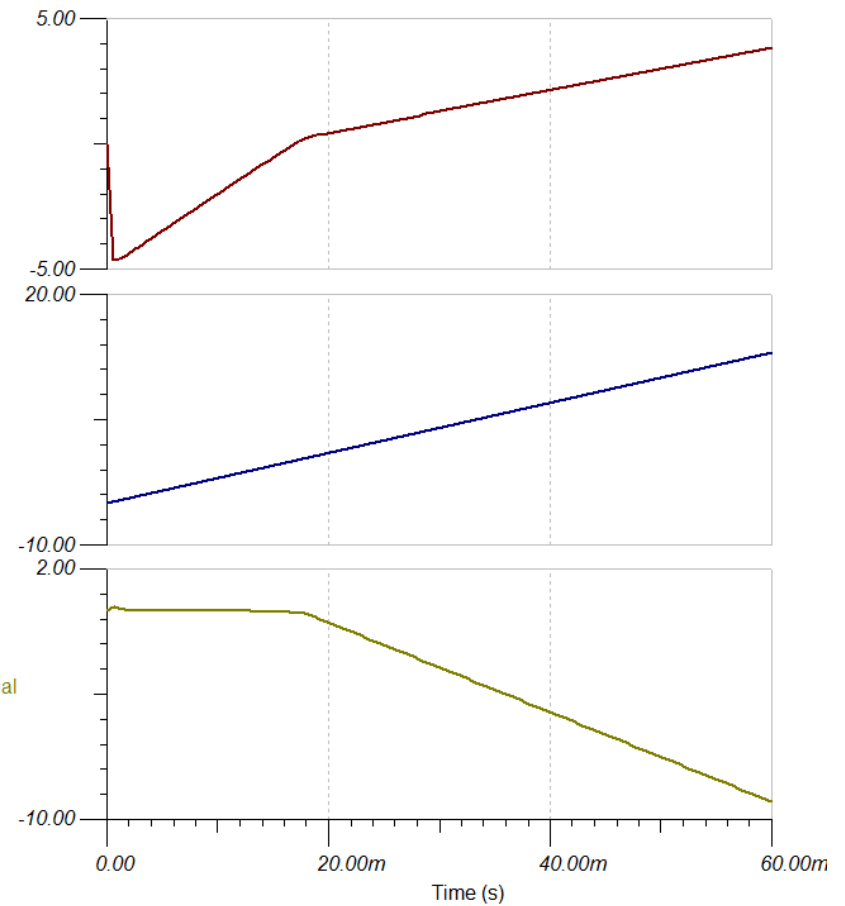
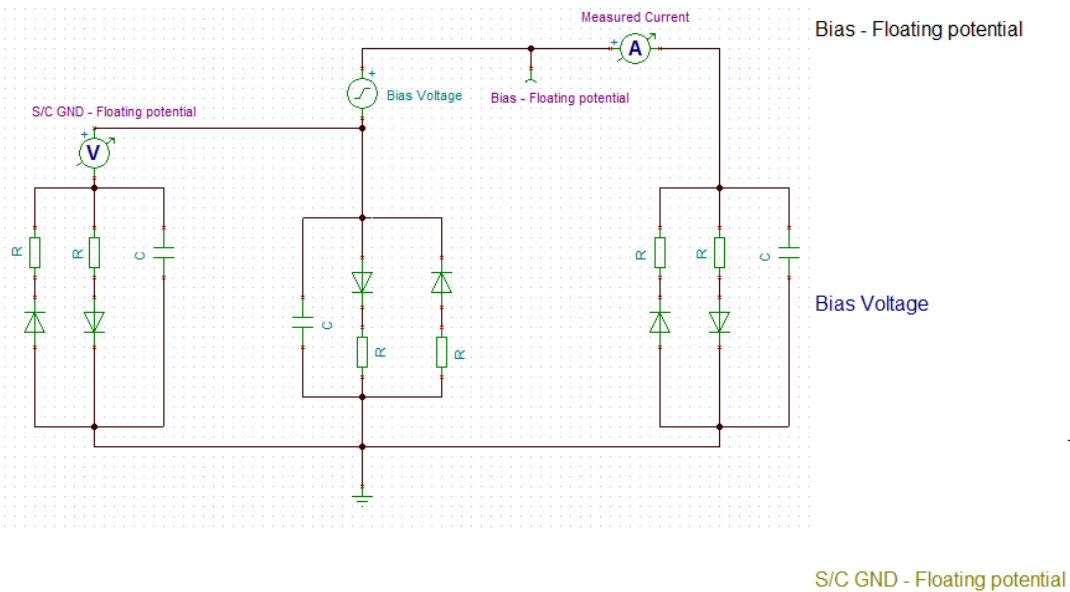


Applied Bias: 6.5 V



Simulations

Electrical circuit modelling and simulation



R and C values derived from measurements in plasma chamber



Probe potential for extreme cases

Maximum probe potential with respect to plasma potential

SPIS simulations for extreme cases

Most unfavorable case:

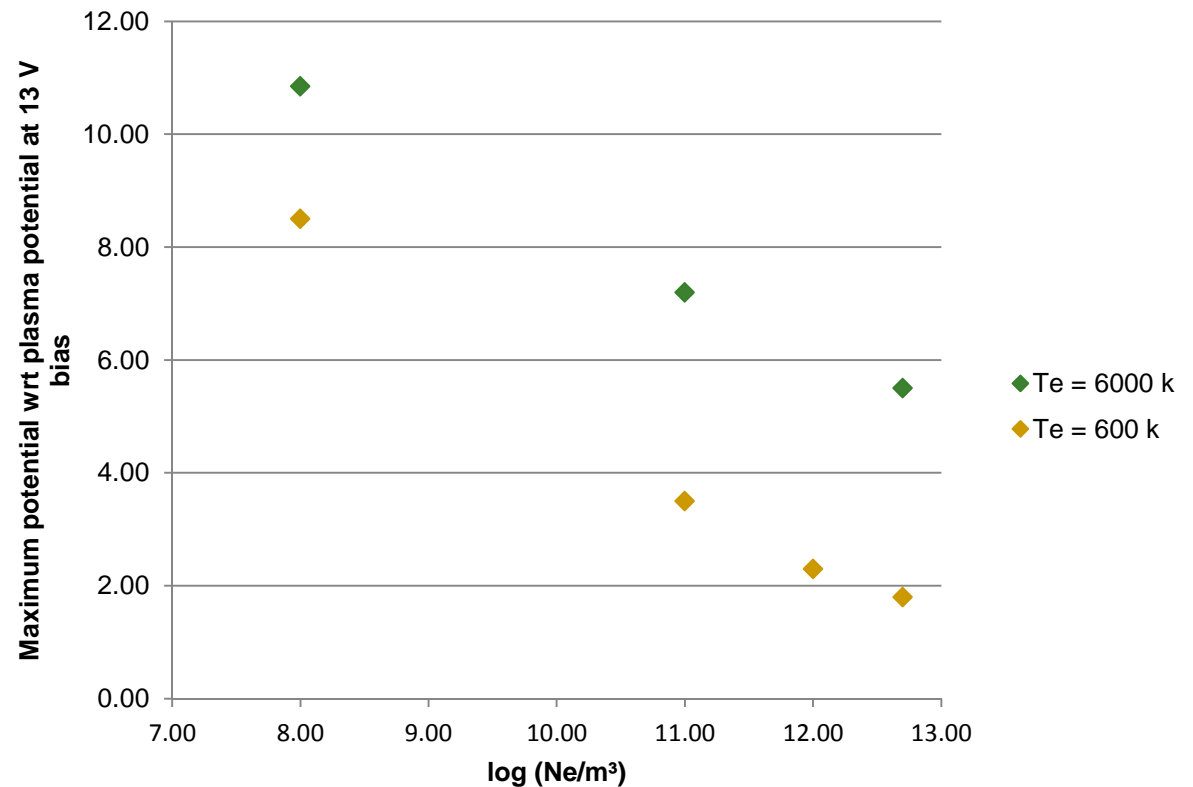
- High Ne
- Low Te
- Eclipse (no photoelectron)

Floating potential:

-0.16 V for Te = 600 K

-1.9 V for Te = 6000 K

=> Always possible to reach electron saturation region





Problem of using LP on board pico-satellite:

Limited conducting area of the S/C with respect to the area of the probe

⇒ **Spacecraft charging** (e- saturation region)

⇒ Drift of the instrument's electrical ground during the measurement

⇒ Unusable data

Risk:

Too low S/C potential: unable to sweep appropriate potentials (e- saturation region)

Proposed solution

- Maximize conducting surface of the S/C
- Measure the floating potential of one probe while measuring the I-V curve with another probe

Allows sweeping bias in e- saturation region even in the worst conditions of the PICASSO mission



Thank you for your attention!

